## FTR - Flight Test Report

Manufacturer	Skywalk GmbH & Co.KG		EAPR-GS-0645/17
	Skywalk GmbH & Co.KG Windeckstr. 4 D-83250 Maquartstein	serial number	
Model	Xalps 3 XS	Location	Stubaital
Comment			



Rev. 2.3 - 15.09.2015 EAPR GmbH - Marktstr. 11 D-87730 Bad Grönenbach - Germany

Date of testing 23.05.2017		
Testpilot	Pascal Purin	
Harness	EAPR Equipment	
Pilot's take off weight	90 kg 70 kg - 90 kg	A STATE OF THE STA
	D (1) (1)	

Range of take off weight

Classification	D
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Nachprüfung

Test-criteria		Evaluation
24. Remarks of testpilot:		
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	Skywalk GmbH & Co.KG Windeckstr. 4 D-83250 Maquartstein	Lo
Model	Xalps 3 XS	
Testpilot	Pascal Purin	
Harness	EAPR Equipment	
Pilot's take off	90	

Date	23.05.2017
Location	Stubaital



presented by

EAPR GmbH- Marktstr. 11 - D-87730 Bad Grönenbach - Germany



Test-criteria		Evaluation	
1. Inflation / take-off - 4.4.1			
Rising behavior		Easy rising, some pilot correction is required	В
Special take off technique required		No	Α
2. Landing - 4.4.2			
Special landing technique required		No	А
3. Speeds in straight flight - 4.4.3			
Trim speed more than 30km/h		Yes	А
•	10km/h	Yes	A
Speed range using the controls larger than 10km/h Minimum speed		25 km/h to 30 km/h	В
4. Control movement - 4.4.4		20 (41)/11 (0 00 (41)/11	
Max. weight in flight up to 80kg			
Max. weight in flight 80 to 100kg		Increasing 35cm - 45cm	D
Max. weight in flight greater than 100kg		increasing South - 45cm	
5. Pitch stability exiting accelerated flig	bt - 4.4.5		-
	4.4.3	Dive forward less than 30°	А
Dive forward angle on exit			
Collapse occurs	ing popularsts	No	A
6. Pitch stability operating controls dur	ing accelerated		
Collapse occurs		No	A
7. Roll stability and damping - 4.4.7			
Oscillations		Reducing	А
8. Stability in gentle spirals - 4.4.8		T	
Tendency to return to straight flight		Spontaneous exit	А
9. Behaviour exiting a fully developed s	spiral dive - 4.4		
Initial response of glider (first 180°)		No immediate reaction	В
Tendency to return to straight flight		Spontaneous exit	Α
Turn angle to recover normal flight		1080° to 1440°, spontaneous recovery	С
10. Symmetric front collapse - 4.4.10			
Folding lines used		No	
Entry	30%	Rocking back less than 45°	Α
Recovery	trim speed ~ 30%	Spontaneous in 3 to 5 sec	В
Dive forward angle on exit	spec	Dive forward 30° - 60° Entering a turn of less than 90°	В
Cascade occurs	trim	No	Α
Entry	%09	Rocking back less than 45°	Α
Recovery	, D	Spontaneous in 3 to 5 sec	В
Dive forward angle on exit	trim speed > 50%	Dive forward 60° - 90° Entering a turn of less than 90°	D
Cascade occurs	trim	No	Α
Entry	20%		-
Recovery	<u>^</u>		-
Dive forward angle on exit	accelerated		-
Cascade occurs	ассе		-
11. Exiting deep stall (parachutal stall)	- 4.4.11		
Deep stall achieved		Yes	
Recovery		Spontaneous in less than 3 sec	А
Dive forward angle on exit		30° - 60°	В
Change of course		Changing course less than 45°	Α
Cascade occurs		No	А
12. High angle of attack recovery - 4.4.	12	<u> </u>	
Recovery			-
Cascade occurs			_
Caccade coours			
13. Recovery from a developed full stal	Ι . Δ Δ 13		

Collapse			-
Cascade occurs (other than collapse)			-
Rocking backward			-
Line tension			-
14. Asymmetric collapse (trim speed) -	- 4.4.14		
Folding lines used		No	
Change of course until re-inflation	8		-
Re-inflation behavior	laps		-
Total change of course	oo %		-
Collapse on the opposite side occurs	trim speed, max 50% collapse		-
Twist occurs	a e		-
Cascade occurs			-
Change of course until re-inflation	es S	90° - 180° Dive or roll angle 60° - 90°	D
Re-inflation behavior	trim speed, max 75% collapse	Inflates in less than 3 sec from start of pilot action	С
Total change of course	trim speed, x 75% colla	Less than 360°	A
Collapse on the opposite side occurs	trim × 75	No	A
Twist occurs	E E	No	A
Change of source until relighting		No	А
Change of course until re-inflation  Re-inflation behavior	es		-
	ited, ollap		-
Total change of course	accelerated, x 50% collap		-
Collapse on the opposite side occurs Twist occurs	accelerated, max 50% collapse		-
Cascade occurs	in a		-
Change of course until re-inflation			-
Re-inflation behavior	ese		-
Total change of course	accelerated, tx 75% collap		-
Collapse on the opposite side occurs	elera % c		-
Twist occurs	accelerated, max 75% collapse		
Cascade occurs	Ĕ		_
15. Directional control with a maintaine	ed asymmetric c	ollanse - 4.4.15	
Able to keep course straight		Yes	А
180° turn away from the collapsed side po	ossible in 10 sec	Yes	A
Amount of control range between turn and		25% to 50% of the symmetric control travel	C
16. Trim speed spin tendency - 4.4.16			
Spin occurs			-
17. Low speed spin tendency - 4.4.17		l	
Spin occurs		No	Α
18. Recovery from a developed spin -	4.4.18		
Spin rotation angle after release			-
Cascade occurs			-
19. B-line-stall - 4.4.19			
Change of course before release			NA
Behaviour before release			NA
Recovery			NA
Dive forward angle on exit			NA
Cascade occurs			NA
20. Big ears - 4.4.20			
Entry procedure			-
Behaviour during big ears			-
Recovery			-
Dive forward angle on exit			-
21. Big Ears in accelerated flight - 4.4.2	21		
Entry procedure			-
Behaviour during big ears			-
Recovery			-
Dive forward angle on exit			-
Behaviour immediately after releasing the accelarator while m			-
23. Alternative means of directional co	ntrol - 4.4.22	Lv	
180° turn achievable in 20 sec		Yes	A
Stall or spin occurs		No	А
23. Any other flight procedure and/or c	ontiguration des	scribed in the user's manual - 4.4.23	<b>N</b> 10
Procedure works as descibed			NA
			N.I.A
Procedure suitable for novice pilots Cascade occurs			NA NA